

NASA TECHNICAL NOTE



NASA TN D-6877

2.1



NASA TN D-6877

LOAN COPY: RETURN TO
AFWL (DOUL)
KIRTLAND AFB, N. M.

ADDITION-TYPE POLYIMIDES FROM SOLUTIONS OF MONOMERIC REACTANTS

*by Peter Delvigs, Tito T. Serafini,
and George R. Lightsey*

*Lewis Research Center
Cleveland, Ohio 44135*



0133722

1. Report No. NASA TN D-6877	2. Government Accession No.	3. Recipient's Catalog No.	
4. Title and Subtitle ADDITION-TYPE POLYIMIDES FROM SOLUTIONS OF MONOMERIC REACTANTS		5. Report Date August 1972	
		6. Performing Organization Code	
7. Author(s) Peter Delvigs, Tito T. Serafini, and George R. Lightsey		8. Performing Organization Report No. E-6750	
		10. Work Unit No. 134-03	
9. Performing Organization Name and Address Lewis Research Center National Aeronautics and Space Administration Cleveland, Ohio 44135		11. Contract or Grant No.	
		13. Type of Report and Period Covered Technical Note	
12. Sponsoring Agency Name and Address National Aeronautics and Space Administration Washington, D. C. 20546		14. Sponsoring Agency Code	
15. Supplementary Notes			
16. Abstract <p>The monomeric reactants approach was used to fabricate addition-type polyimide/graphite fiber composites with improved mechanical properties and thermal stability characteristics over those of composites derived from addition-type amide acid prepolymers. A screening study of 24 different monomer combinations was performed. The results of a more extensive investigation of a selected number of monomer combinations showed that the combination providing the best thermomechanical properties was 5-norbornene-2,3-dicarboxylic acid monomethyl ester/4,4'-methylenedianiline/3,3',4,4'-benzophenonetetracarboxylic acid dimethyl ester at a molar ratio of 2/3.09/2.09.</p>			
17. Key Words (Suggested by Author(s)) Thermally stable polymers Monomeric reactants Polyimides Fiber reinforced polymers		18. Distribution Statement Unclassified - unlimited	
19. Security Classif. (of this report) Unclassified	20. Security Classif. (of this page) Unclassified	21. No. of Pages 20	22. Price* \$3.00

ADDITION-TYPE POLYIMIDES FROM SOLUTIONS OF MONOMERIC REACTANTS

by Peter Delvigs, Tito T. Serafini, and George R. Lightsey

Lewis Research Center

SUMMARY

The purpose of this investigation was to fabricate addition-type polyimide/graphite fiber composites with improved mechanical properties and thermal stability characteristics over those of P10P/graphite fiber composites. In this study a monomeric reactants approach was used, where the fibers were impregnated with solutions of appropriate monomers instead of an amide acid prepolymer solution. The combination of monomers used for preparing the prepreg solutions was (1) the monomethyl ester of 5-norbornene-2,3-dicarboxylic acid (NE), (2) the dimethyl ester of pyromellitic acid (PMDE) or the dimethyl ester of 3,3',4,4'-benzophenonetetracarboxylic acid (BTDE), and (3) an aromatic diamine.

This report describes a screening study of 24 polyimide/HTS graphite fiber composites (1.27 by 6.35 cm (1/2 by 2 $\frac{1}{2}$ in.)) fabricated from monomeric solutions differing in chemical composition or stoichiometry. Based on initial interlaminar shear strength and isothermal weight loss in air after 600 hours at 315° C (600° F), several monomer combinations were selected for a more detailed evaluation. Large composite panels (7.62 by 25.4 cm (3 by 10 in.)) were fabricated from the selected monomer combinations, and their mechanical properties and thermal stability characteristics determined at 315° C (600° F) over a period of 600 hours.

Based on the results of this investigation, the most suitable aromatic diamine was found to be 4,4'-methylenedianiline (MDA). The monomer composition NE/MDA/BTDE at three different stoichiometric ratios yielded composites that possessed long-term mechanical properties and a thermooxidative stability superior to those of P10P composites. The composites fabricated from the NE/MDA/BTDE formulation at a molar ratio of 2/3.09/2.09 gave the best overall retention of mechanical properties of all panels studied. The interlaminar shear strength of this composite was 45 percent higher than that of P10P composites after a 600-hour exposure in air at 315° C (600° F), and the flexural strength was 75 percent higher than that of P10P composites.

INTRODUCTION

One of the recent breakthroughs in the chemistry of high temperature resins is the development of a novel class of polyimides which cure by an addition reaction (refs. 1 and 2). This approach uses low molecular weight amide acid prepolymers end capped with reactive alicyclic rings that polymerize into thermally stable polyimides without the evolution of byproducts. A commercially available resin of this type is P13N. Another version, known as P10P, having considerably improved thermooxidative stability, has also been developed (ref. 3). From our inhouse studies of matrix resins, we have observed that P10P prepolymer solutions exhibit a very limited shelf life at room temperature, as evidenced by precipitation or gel formation. Prior to the appearance of a precipitate or gel there is a steady decrease of solution viscosity.

To circumvent these limitations, we have demonstrated that it is not necessary to synthesize end-capped amide acid prepolymers. Instead, graphite fibers are impregnated with solutions of appropriate monomers, and the end-capped prepolymer is formed in situ by the application of heat (ref. 4). Solutions of monomeric reactants offer several advantages over amide acid prepolymer solutions, including excellent shelf life, increased solids loading, and lower viscosity. In addition, relatively nontoxic, low-boiling alcoholic solvents such as methanol can be used as solvents for the monomers.

The purpose of the present study was, by using the monomeric reactants approach, to fabricate polyimide/graphite fiber composites with improved mechanical properties and thermal stability characteristics over those of P10P/graphite composites.

This report describes a screening study of 24 polyimide/graphite fiber composites (1.27 by 6.35 cm (1/2 by 2 $\frac{1}{2}$ in.)) fabricated from monomeric solutions different in chemical composition or stoichiometry. The combination of monomers used for preparing the prepreg solutions was (1) the monomethyl ester of 5-norbornene-2,3-dicarboxylic acid, (2) the dimethyl ester of pyromellitic acid or 3,3',4,4'-benzophenonetetracarboxylic acid, and (3) an aromatic diamine.

Based on the screening study results, 7.62 by 25.4 centimeter (3 by 10 in.) composites were fabricated from several selected monomer solutions. Their mechanical properties and thermal stability characteristics were determined at 315° C (600° F) over a period of 600 hours.

EXPERIMENTAL PROCEDURE

Monomers

The monomers used in this study are shown in table I. The monomethyl ester of

TABLE I. - MONOMERS USED FOR POLYIMIDE SYNTHESIS

Structure	Name	Abbreviation
	Monomethyl ester of 5-norbornene-2,3-dicarboxylic acid	NE
	Dimethyl ester of 3,3',4,4'-benzophenone-tetracarboxylic acid	BTDE
	2,5-Dicarbomethoxyterephthalic acid	PMDE
	4,4'-Methylenedianiline	MDA
	4,4'-Oxydianiline	ODA
	4,4'-Thiodianiline	TDA
	4,4'-Sulfonyldianiline	SDA
	Benzidine	BZD
	4,4'-Diaminostilbene	DAS
	<u>p</u> -Phenylenediamine	PPDA
	<u>m</u> -Phenylenediamine	MPDA

5-norbornene-2,3-dicarboxylic acid (NE) was prepared according to the method of Walton (ref. 5). The 2,5-dicarbomethoxyterephthalic acid (PMDE) was prepared according to Bell and Jewell (ref. 6).

The dimethyl ester of 3,3',4,4'-benzophenonetetracarboxylic acid (BTDE) was synthesized by refluxing a suspension of 3,3',4,4'-benzophenonetetracarboxylic dianhydride (96.6 g, 3.00×10^{-1} mole) in anhydrous methanol (95 ml) until the solid had dissolved (2 to 3 hr) and then for an additional hour. Excess methanol was distilled off under reduced pressure at 50° C. The brownish oil was dried further in vacuum (10^{-3} torr) at room temperature for 24 hours to give 115.2 grams (99-percent yield) of the isomer mixture BTDE as a crystalline solid. The solid crystals are hygroscopic and were therefore stored in a sealed container under nitrogen.

The 4,4'-methylenedianiline (MDA), 4,4'-oxydianiline (ODA), 4,4'-thiodianiline (TDA), and 4,4'-sulfonyldianiline (SDA) were obtained from a commercial source and recrystallized from 2-propanol. Both p-phenylenediamine (PPDA) and m-phenylenediamine (MPDA) were used as received. Benzidine (BZD) was prepared from its dihydrochloride salt by treatment with dilute aqueous KOH. The crude BZD was recrystallized from a mixture of water and acetone (2:1 by volume). The 4,4'-diaminostilbene (DAS) was prepared from its dihydrochloride salt by treatment with dilute aqueous KOH. The crude DAS was recrystallized from 95 percent ethanol.

Prepreg Solutions

The solvent used was anhydrous methanol unless otherwise indicated (tables II and III). The solutions were prepared at a solids loading of 50 weight percent. The stoichiometry of the reactants was governed by the desired number of imide rings or by the desired percentage of alicyclic content.

Composite Fabrication and Testing

Prepreg tapes were made by drum winding and impregnating Hercules HTS graphite fiber with the various solutions of monomeric reactants to yield prepregs that contained 40 percent monomers and 60 percent fiber by weight. The prepregs made from monomer solutions in methanol were heated on the drum for 2 hours at 50° C (122° F) to reduce the solvent content to less than 10 percent by weight. The prepregs made from monomer solutions in 1-methyl-2-pyrrolidinone were heated for 2 hours at 50° C (122° F) and then for 10 minutes at 120° C (248° F) to effect a similar solvent content reduction. After this treatment, the prepreg tapes were flexible, but not tacky.

TABLE II. - PROPERTIES OF VARIOUS 1.27 BY 6.35 CM (1/2 BY 2 1/2 IN.) POLYIMIDE/HTS

GRAPHITE FIBER COMPOSITES

Sample	Monomeric solution composition	n	FMW	Processing conditions		Laminate properties				Composite weight loss, percent ^a
				Imidization time at 205 ^o C (400 ^o F), hr	Dwell time at 315 ^o C (600 ^o F), sec	Interlaminar shear at 24 ^o C (75 ^o F)		Interlaminar shear at 315 ^o C (600 ^o F)		
						N/m ²	ksi	N/m ²	ksi	
b ₁	NE/MDA/BTDE	2.087	1500	3	50	105.5×10 ⁶	15.3	41.4×10 ⁶	6.0	9.5
2	↓	2.603	1750	↓	40	99.3	14.4	37.9	5.5	9.8
3	↓	2.913	1900	↓	40	103.4	15.0	35.2	5.1	10.0
4	↓	3.120	2000	↓	30	67.6	9.8	31.7	4.6	11.4
5	↓	5.186	3000	↓	30	56.5	8.2	35.2	5.1	28.0
b ₆	NE/MDA/PMDE	1.342	1000	2	50	74.5	10.8	42.7	6.2	17.6
7	↓	1.868	1200	↓	40	84.8	12.3	42.7	6.2	17.9
8	↓	2.658	1500	↓	30	42.1	6.1	20.0	2.9	20.8
9	NE/MDA/ (1BTDE:1PMDE)	2.338	1500	↓	50	98.6	14.3	42.7	6.2	10.6
10	↓	2.917	1750	3	40	93.8	13.6	37.9	5.5	8.6
11	↓	3.264	1900	↓	↓	86.9	12.6	31.0	4.5	10.9
12	NE/MPDA/BTDE	2.793	1500	↓	↓	87.6	12.7	31.0	4.5	17.3
13	NE/MPDA/BTDE	3.426	1750	4	↓	56.3	8.2	31.0	4.5	16.3
14	NE/ODA/BTDE ^c	2.074	1500	3	↓	62.1	9.0	33.1	4.8	24.2
15	NE/ODA/BTDE ^c	2.897	1900	3	30	69.0	10.0	30.3	4.4	19.1
16	NE/TDA/PMDE	2.492	1500	2	50	51.0	7.4	25.5	3.7	22.6
17	NE/TDA/BTDE	1.975	1500	3	40	46.9	6.8	25.5	3.7	14.6
18	NE/SDA/PMDE ^c	1.070	1000	2	50	23.4	3.4	17.9	2.6	15.6
19	NE/SDA/BTDE ^c	1.000	1074	2	50	31.7	4.6	17.9	2.6	15.5

^aAfter 600 hours in air at 315° C (600° F).^bData taken from ref. 4.^cSolvent - 1-methyl-2-pyrrolidinone.

TABLE III. - UNSUCCESSFUL MONOMER COMBINATIONS

Sample	Monomeric solution composition	n	FMW
^a 20	NE/PPDA/PMDE	2.069	1000
21	NE/PPDA/BTDE	1.000	794
22	NE/BZD/BTDE	2.179	1500
^a 23	NE/DAS/BTDE	1.004	1000
^a 24	NE/DAS/BTDE	2.012	1500

^aSolvent - 1-methyl-2-pyrrolidinone.

Unidirectional composites were molded essentially according to the method of Burns, et al. (ref. 3). Twelve plies of prepreg 1.27 by 6.35 centimeters ($1/2$ by $2\frac{1}{2}$ in.) were stacked between aluminum foil and heated for 2 to 4 hours in 205°C (400°F) oven. The imidized prepreps were then placed in a mold preheated to 315°C (600°F). The mold was closed, contact pressure (dwell time) was applied for 30 to 50 seconds, and then a pressure of 6.24×10^6 newtons per square meter (906 psi) was applied. After 30 minutes at these conditions the pressure was released and the mold was allowed to cool slowly to room temperature. The imidization time at 205°C (400°F) and dwell time at 315°C (600°F) for each of the small composites made are given in table II. The processing conditions for the larger panels, 7.62 by 25.4 centimeters (3 by 10 in.), differed from those previously described in the following respects. The stacked prepreg was placed between two three-ply layers of Teflon-coated glass fabric bleeder cloth and a 4.1-kilogram mass (9-lb) was placed on the stack during imidization. After imidization, the layers of bleeder cloth were removed, the stack was placed between aluminum foil and molded at 3.85×10^6 newtons per square meter and 315°C (425 psi and 600°F). The larger panels had a resin content of 34 to 36 weight percent.

Flexural tests were made on a three-point loading fixture with a fixed span of 5.08 centimeters (2 in.). The specimens were 0.635 centimeter wide by 6.03 centimeters long ($1/4$ in. wide by $2\frac{3}{8}$ in. long). The nominal thicknesses of the small and large panels were 0.23 and 0.25 centimeter (0.090 and 0.100 in.), respectively. The rate of center loading was 0.127 centimeter per minute (0.05 in./min). The interlaminar shear tests were conducted at a constant span to thickness ratio of 5 using the interlaminar shear test fixture described by Hanson and Serafini (ref. 7). The 315°C (600°F) flexural and interlaminar shear tests were performed in an environmental heating chamber. Load was applied to the specimens after a 15-minute soak at 315°C (600°F). Forced convection air ovens were used for the long-term isothermal exposures at 315°C (600°F). The air change rate was 100 cubic centimeters per minute ($6.1\text{ in.}^3/\text{min}$).

The data presented in table II are averages of two or more measurements at each condition. Six or more tests at each condition were averaged to give the data shown in figures 2 to 5, 8, and 10 to 16.

RESULTS AND DISCUSSION

Composite Screening Study

In a prior investigation (ref. 4) we showed that the method of preparing composites from solutions of monomeric reactants provides composites with properties comparable

to those of composites made from amide acid prepolymers. In addition, solutions of monomeric reactants offer several advantages over amide acid prepolymer solutions: most important, low-boiling, relatively nontoxic alcohols such as methanol can be used as solvents instead of high-boiling, toxic solvents such as N,N-dimethylformamide.

In our previous work (ref. 4) we noted that certain composites made from monomer solutions containing BTDE had unexpectedly high thermooxidative stability. We therefore undertook a more detailed investigation to establish the optimum molar ratio of the reactants (NE/MDA/BTDE).

Another objective of this investigation was to determine if the alicyclic content of the NE/MDA/PMDE system could be lowered further to improve its thermooxidative stability.

In our previous work (ref. 4) the number of diamines used in the prepreg solutions was limited to either 4,4'-methylenedianiline or 4,4'-thiodianiline. In this investigation six additional diamines were used in a composite screening study.

Also discussed in a comparison of composite properties from monomer solutions where the alicyclic content was kept constant, but the number of imide rings was varied.

The chemical composition, some curing parameters, and some properties of 19 various 1.27 by 6.35 centimeter ($1/2$ by $2\frac{1}{2}$ in.) composites are summarized in table II. The stoichiometry of the reactants was governed by the desired number of imide rings or by the desired percentage of alicyclic content. The term formulated molecular weight (FMW) is used as a convenience throughout this discussion. The FMW is considered to be the average molecular weight of the imidized prepolymer that could have been formed if amide acid prepolymers had been synthesized. For example, the structure of the imidized prepolymer resulting from monomeric solution 1 (table II) is expected to be similar to the structure shown in figure 1. For any given monomer combination, the molar ratio of 5-norbornene-2,3-dicarboxylic acid monomethyl ester/diamine/tetracarboxylic acid dimethyl ester was $2/(n+1)/n$.

The dwell time (contact pressure time) of the composites in the heated mold was varied from 30 to 50 seconds. A decrease of dwell time tends to increase the amount of resin flow by minimizing the number of crosslinks and chain extensions formed prior to application of pressure. It should be pointed out that the composite processing conditions used throughout this investigation were essentially those which had been optimized for amide acid prepolymer solutions (ref. 3).

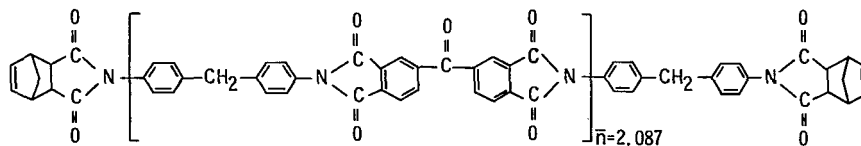


Figure 1. - Imidized prepolymer from NE/MDA/BTDE. Formulated molecular weight, FMW = 1500.

Samples 1 to 5 are chemically similar to the commercially available P13N resin. All of them, however, have a lower alicyclic content than P13N. Samples 1 to 3 possessed a combination of unusually high thermooxidative stability and high initial interlaminar shear at room temperature. They were therefore chosen for a more detailed study of mechanical properties after exposure at 315° C (600° F). Samples 4 and 5 were excluded from further study because of considerably inferior interlaminar shear values at room temperature.

Sample 6 has a composition equivalent to that of P10P prepolymer. Its properties were shown to be virtually identical with those of composites made from P10P. Samples 7 and 8 have the same chemical composition as sample 6, but differ in stoichiometry. Sample 7 was selected for further study because of its high interlaminar shear at 315° C (600° F). Sample 8, having significantly lower interlaminar shear at room temperature and 315° C (600° F), was excluded from further study.

The monomer combination used in samples 9 to 11 was selected to investigate the effect of increasing the number of imide linkages over that in the corresponding samples 1 to 3 while maintaining a constant alicyclic content. For example, sample 9 has the same alicyclic content as sample 1, but a higher number of imide linkages. The higher number of imide linkages is achieved by partial substitution of PMDE for BTDE. Theoretically this should give a correspondingly higher thermooxidative stability. Actually, the results indicate that the thermooxidative stability of samples 9 to 11 is approximately equivalent to that of samples 1 to 3. Sample 9 was selected for further evaluation as a representative of the NE/MDA/(1PMDE:1BTDE) monomer combination.

The monomer combinations in samples 12 to 19 (table II) and 20 to 24 (table III) were selected to determine the effect of seven additional diamine constituents on composite properties. These samples contained BTDE as the tetracarboxylic acid dimethyl ester component, with the exception of samples 16, 18, and 20 where PMDE was used in an effort to increase the number of imide linkages and therefore thermal stability. Sample 12 displayed excellent resin flow, but there was some blistering on the surface of the composite. Because the blisters could not be eliminated after numerous attempts, sample 12 was excluded from further study. Samples 13 to 19 were eliminated from advanced composite evaluation because of either a high composite weight loss or low initial interlaminar shear values. Composites from samples 20 to 24 (table III) had very poor consolidation and delaminated badly when subjected to mechanical testing. Therefore, it was impossible to obtain meaningful mechanical property data, and these samples were not subjected to further investigation.

Advanced Composite Evaluation

Composite properties of samples 1 to 3 after exposure at 315° C (600° F) as a function of time are shown in figures 2 to 5. These tests were performed on 12-ply composites, 7.62 by 25.4 centimeters (3 by 10 in.). For comparison purposes, composite data for sample 6 are also presented in figures 2 to 5. Sample 6 is chemically equivalent to the composites prepared from the P10P prepolymer (developed by Burns, et al., ref. 3), which had yielded the best thermooxidative stability for addition-type polyimides up to the present investigation.

Figure 2 shows that the composite weight loss for samples 1 to 3 was considerably lower than that of sample 6. It is difficult to attach too much significance to the weight

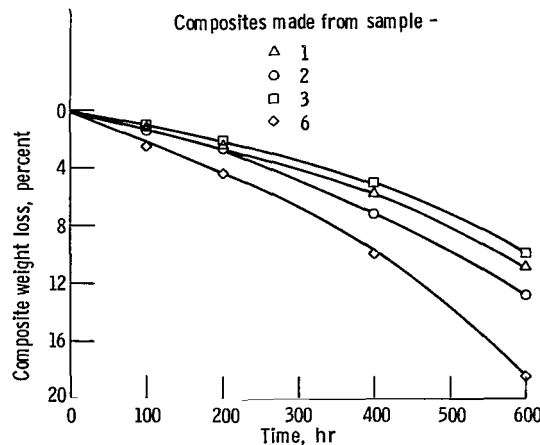


Figure 2. - Percent weight loss of polyimide/HTS graphite fiber composites exposed in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

loss differences among samples 1 to 3 because of the vagaries of isothermal analysis. However, it is significant to note that the average percent weight loss (11.2 percent) for samples 1 to 3 after 600 hours was 39 percent lower than the weight loss for sample 6.

Figures 3 to 5 show the variation of mechanical properties for samples 1 to 3 and 6 after exposure at 315° C (600° F) as a function of exposure time. Also presented are the room temperature and short time 315° C (600° F) properties.

The interlaminar shear strength is shown in figure 3. The range of the room temperature values is 95×10^6 to 123×10^6 newtons per square meter (13 800 to 17 800 psi).

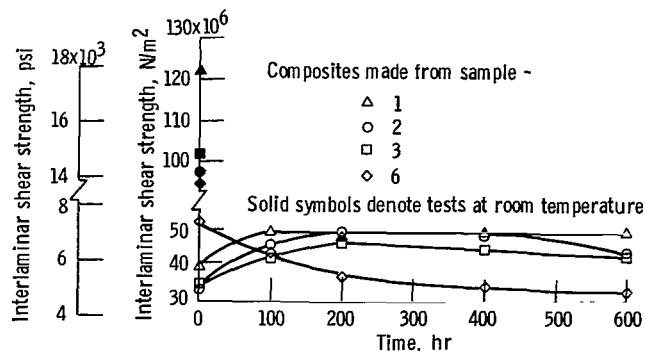


Figure 3. - Interlaminar shear strength of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

These values are somewhat higher than those previously reported (ref. 7) for A-type polyimide/HTS fiber composites. There is a considerable drop in the interlaminar shear strength when the composites are tested at 315° C (600° F), to as low as 34.5 newtons per square meter (5000 psi). There was a greater decrease for samples 1 to 3 than for sample 6. The interlaminar shear strength decrease for sample 6 is nearly identical to that reported by Hanson and Serafini (ref. 7).

It appears that composite samples 1 to 3 underwent a postcure on exposure in air at 315° C (600° F), since the interlaminar shear strength increased during exposure, whereas that of sample 6 decreased with time. The important fact to note is that the interlaminar shear strength of samples 1 to 3 remained nearly constant from 100 to 600 hours. The strength after 600 hours for samples 1 to 3 averaged approximately 45×10^6 newtons per square meter (6500 psi), compared to 33×10^6 newtons per square meter (4800 psi) for sample 6, representing a 35-percent increase. The data for samples 1 to 3 represent the best interlaminar shear retention for A-type polyimide/HTS fiber composites reported to date. The interlaminar shear strength of sample 1 increased to 48.5×10^6 newtons per square meter (7000 psi) after 200 hours of exposure and remained constant up to 600 hours of exposure. It appears that the interlaminar shear strength would have remained constant for a considerably longer time if the exposure time had been extended. The interlaminar shear strengths of samples 2 and 3 began to decrease after reaching maximum values after 200 hours to approximately 43×10^6 newtons per square meter (6200 psi) after 600 hours. These results are somewhat contrary to the expected improvement in thermooxidative stability and mechanical properties for samples 2 and 3, resulting from the higher number of imide rings and lower alicyclic content. This discrepancy could possibly be attributed to slightly poorer resin flow during the final curing step for samples 2 and 3. In view of these results we feel that the stoichiometry of reactants used in sample 1 is preferred.

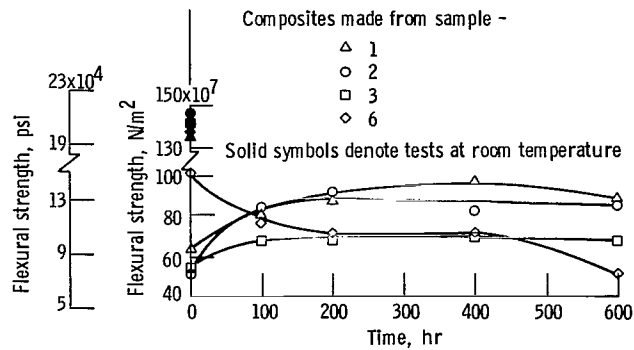


Figure 4. - Flexural strength of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

The flexural strength results are shown in figure 4. Many of the comments presented in the discussion of interlaminar shear strength are also applicable here. For example, a postcure effect is also evident for composite samples 1 to 3. The flexural strength values for sample 3 are lower than those for samples 1 and 2 throughout the 600-hour exposure. This could again be attributed to minor differences in resin flow during processing. It is important to note, however, that after 600 hours of exposure the flexural strength of samples 1 and 2 was nearly twice that of sample 6.

The variation of the flexural modulus of elasticity as a function of exposure time is shown in figure 5. Here, too, a postcure effect was observed. The modulus values for samples 1 to 3 did not differ appreciably, and remained essentially constant from 100 to 600 hours. On the other hand, the modulus of sample 6 showed a marked decrease after 300 hours.

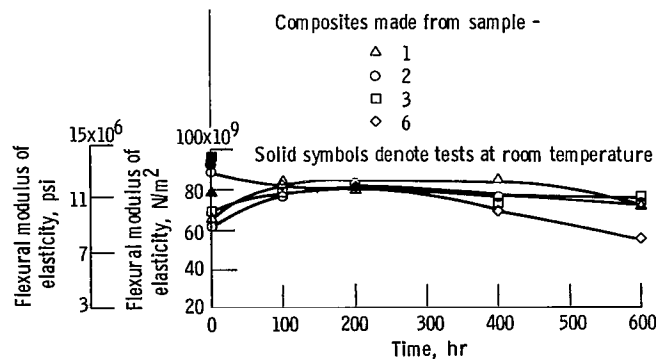
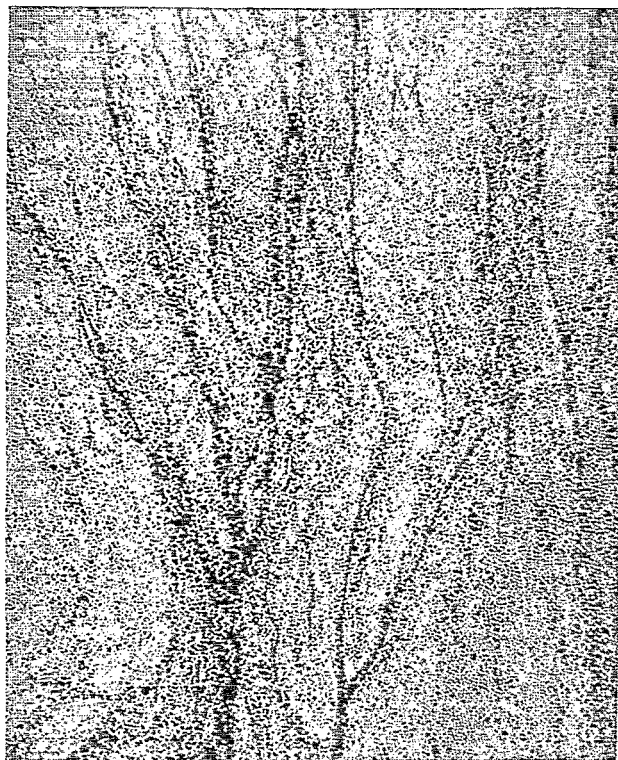
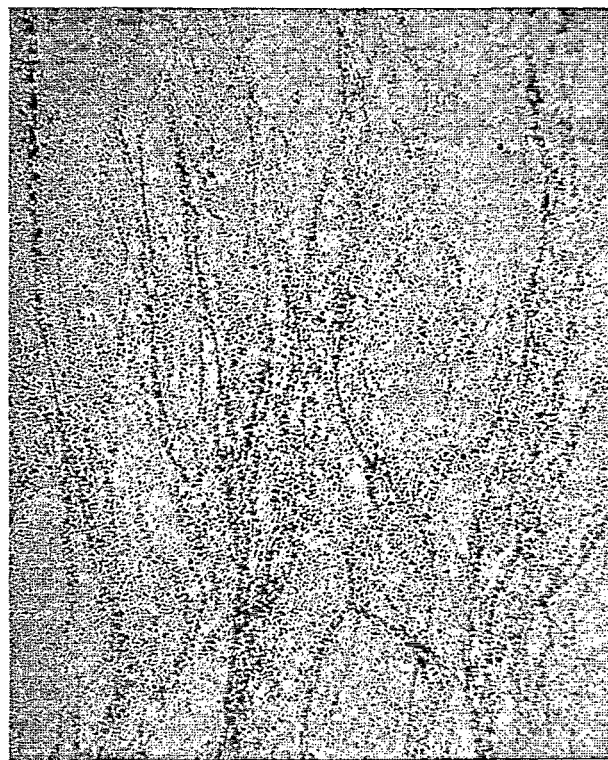


Figure 5. - Flexural modulus of elasticity of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

Photomicrographs of a cross section of composite sample 1, before and after exposure, are shown in figure 6. There is no evidence of voids in the composite as a result of fabrication. After exposure at 315°C (600°F) for 600 hours, the composite shows an extremely small number of voids. This indicates that the original void-free composite was very resistant to oxidative degradation that might occur by a diffusion mechanism. Any degradation that did occur, took place on the surface of the composite. The absence of internal voids after exposure also indicates that the composite is very resistant to thermal decomposition.



(a) Sample 1, unexposed.

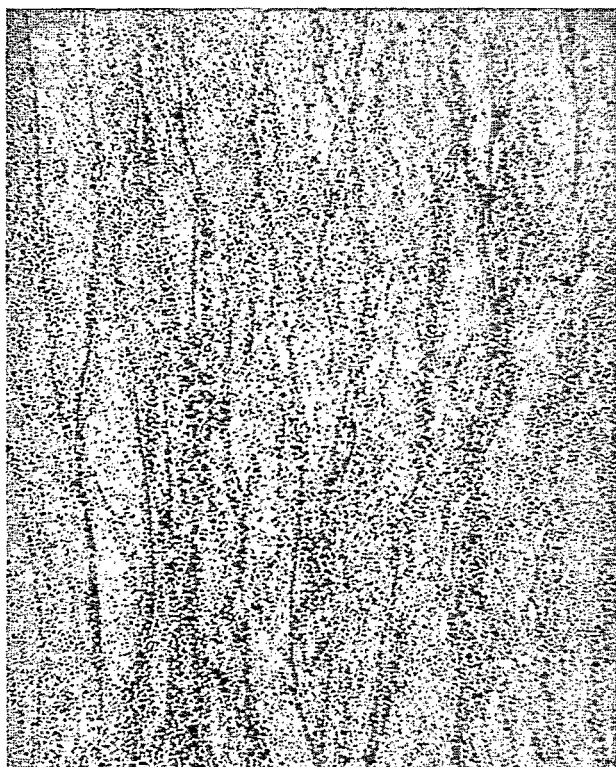


(b) Sample 1, exposed.

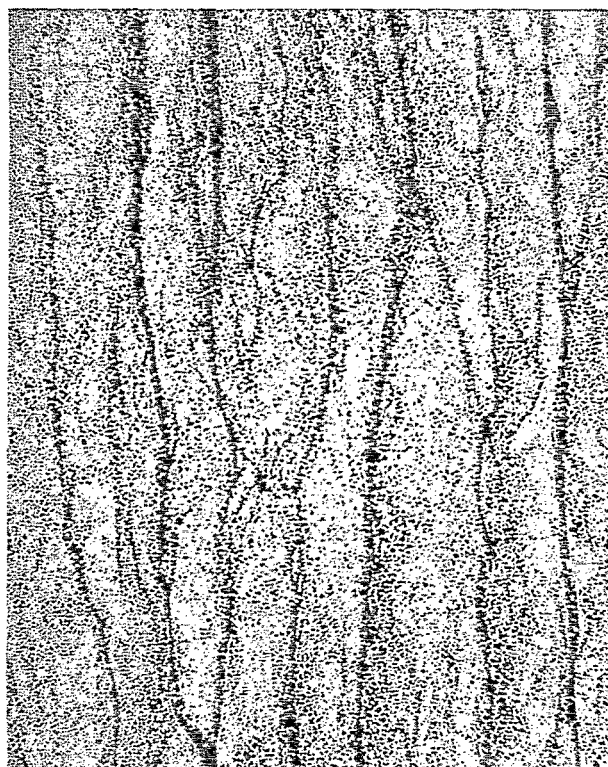
Figure 6. - Photomicrographs of polyimide/HTS graphite fiber composites before and after exposure in air at 315°C (600°F) for 600 hours. X50.

Figure 7 shows the photomicrographs of composite samples 2 and 3 before exposure at 315°C (600°F). The photomicrographs show the presence of a small number of voids in both cases. These voids might be responsible for the slightly inferior mechanical property retention of samples 2 and 3, compared with sample 1.

To sum up the results of this portion of the advanced composite evaluation we conclude that the stoichiometry used in sample 1 provided composites having the best combination of processability and thermomechanical properties. Sample 3, which had the largest number of imide rings and lowest alicyclic content, would be expected to exhibit



(a) Sample 2.



(b) Sample 3.

Figure 7. - Photomicrographs of unexposed polyimide/HTS graphite fiber composites. X50.

the best performance. However, its performance was not as good as that of composite sample 1, which had a fewer number of imide rings. We attribute this finding to the better resin flow and processability obtained using the stoichiometry of sample 1. However, each of the composite samples 1 to 3, containing BTDE as the tetracarboxylic acid component, yielded significantly better property retention than sample 6 (containing PMDE), which had given the best thermomechanical performance up to the present investigation (ref. 4).

Figures 8 to 12 show the effects of long-term exposure at 315°C (600°F) on the composite properties of samples 6 and 7. Sample 7 has the same chemical composition as sample 6, but a higher number of imide rings and a lower alicyclic content.

Composite weight loss data are presented in figure 8. It can be seen that the weight loss experienced by sample 7 after 300 hours at 315°C (600°F) was 17 percent, compared to 7 percent for sample 6. The high weight loss shown by sample 7 is contrary to the results obtained in the screening study (see table II). The reliability of the results was verified by making and testing more than one large panel. We attribute the high composite weight loss to a relatively high void content. These voids might have resulted from a combination of the following two influences:

(1) The higher molar ratio of PMDE in sample 7 prevented adequate resin flow during the final stages of cure.

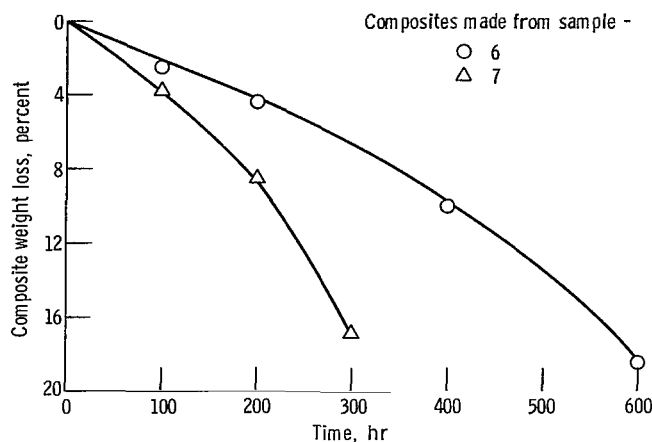


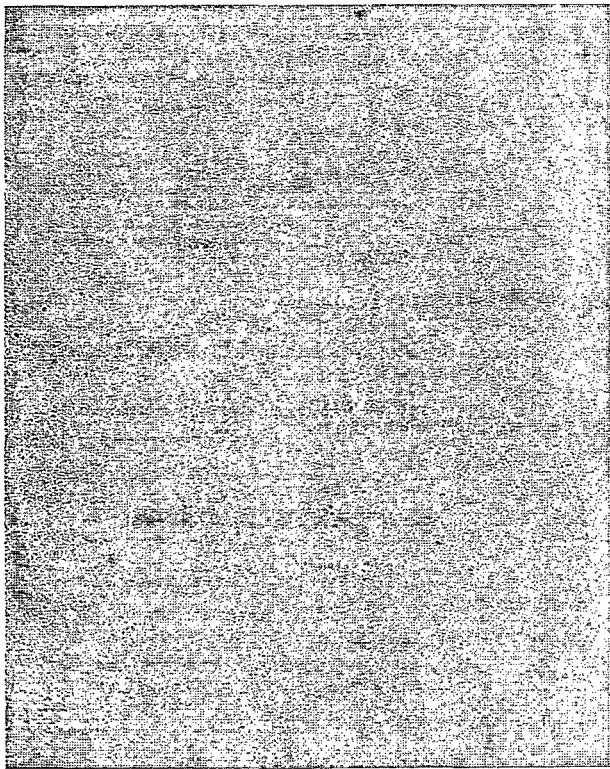
Figure 8. - Percent weight loss of polyimide/HTS graphite fiber composites exposed in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

(2) The other factor which might have contributed to produce a high void content is related to the physical size of the composites.

It is much easier for the byproducts formed during reaction of the monomers and imidization to escape from the preforms for the smaller laminates than for the large laminates. In the large laminates (7.62 by 25.4 cm) some byproducts might not have completely diffused from the preform during the initial staging. These entrapped byproducts were probably volatilized during final curing and could not escape prior to resin consolidation. The photomicrographs of composite samples 6 and 7 are shown in figure 9. It is apparent that sample 7 had a much higher void content, which no doubt contributed to the higher rate of weight loss observed.

The interlaminar shear strength, flexural strength, and flexural modulus of elasticity for samples 6 and 7 as functions of exposure time at 315° C (600° F) are shown in figures 10 to 12. It is obvious that the properties of composite sample 7 are quite inferior to those of sample 6. This finding points out not only the need for thermally stable groups, but more importantly, the need for selecting the proper combination of the three monomer types, as well as stoichiometry. At this point it can be said with considerable assurance that the monomeric reactants approach provides a method of preparing addition-type polyimide composites using BTDE as the tetracarboxylic acid function that are considerably superior to composites made from PMDE.

The remaining sample selected for advanced composite evaluation was sample 9. In sample 9, 50 mole percent of BTDE was replaced with PMDE, because theoretically PMDE should be more stable thermooxidatively than BTDE. In addition, sample 9 possessed the same alicyclic content as sample 1. The photomicrograph of sample 9 was similar to that of sample 1 prior to exposure (fig. 6). The effects of long-term expo-



(a) Sample 6.



(b) Sample 7.

Figure 9. - Photomicrographs of unexposed polyimide/HTS graphite fiber composites. X50.

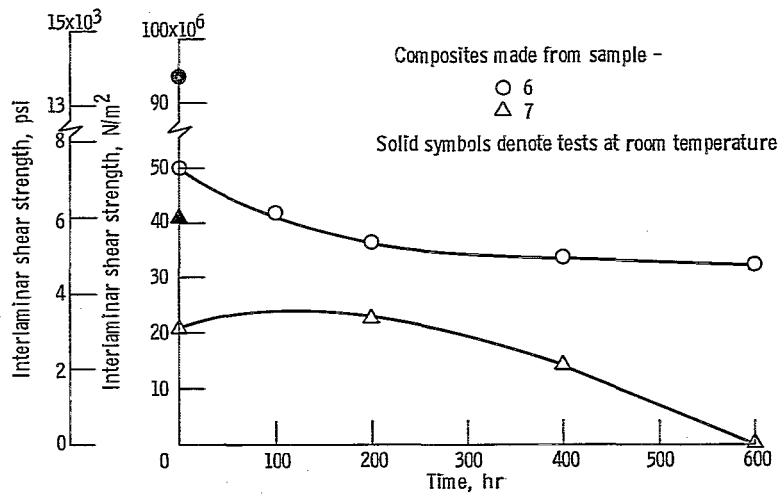


Figure 10. - Interlaminar shear strength of polyimide/HTS graphite fiber composites exposed and tested in air at 315°C (600°F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

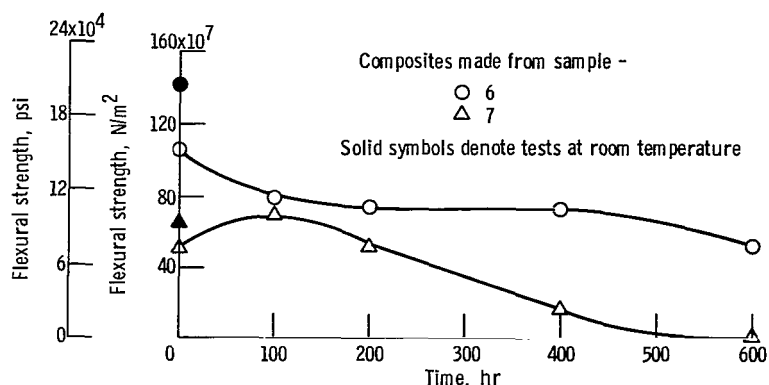


Figure 11. - Flexural strength of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

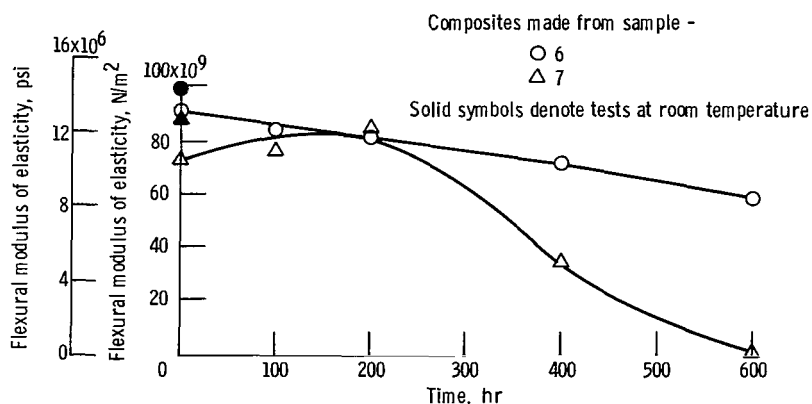


Figure 12. - Flexural modulus of elasticity of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

sure at 315° C (600° F) on the composite properties of sample 9 are shown in figures 13 to 16. For comparison, the properties of sample 1 (containing BTDE) and sample 6 (containing PMDE) are included.

The composite weight loss data presented in figure 13 show that the weight loss curve for sample 9 is almost identical with that for sample 1. It can be seen that partial inclusion of the more stable PMDE did not improve the thermooxidative stability of sample 9 over that of sample 1.

The interlaminar shear strength values are shown in figure 14. The shear strength retention of sample 9 was inferior to that of sample 1. There was not much difference in the interlaminar shear strength of samples 1 and 9 for 200 hours at 315° C (600° F). However, the shear strength of sample 9 began to show a decrease after 200 hours of exposure, and it became nearly identical with that of sample 6 after 600 hours (35 per -

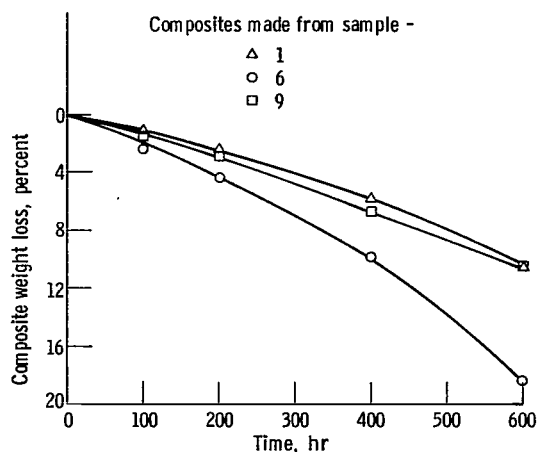


Figure 13. - Percent weight loss of polyimide/HTS graphite fiber composites exposed in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

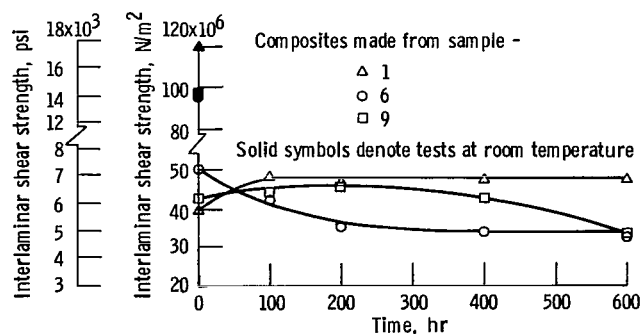


Figure 14. - Interlaminar shear strength of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

cent lower than the shear strength of sample 1).

The variation of flexural strength as a function of exposure time is shown in figure 15. Again it is apparent that the flexural strength of sample 9 was considerably lower than that of sample 1, and only slightly higher than that of sample 6. After exposure for 600 hours the flexural strength of sample 9 was 30 percent lower than that of sample 1.

The variation of the flexural modulus, shown in figure 16, was approximately the same for all three samples.

In summary, the partial replacement of BTDE with PMDE did not significantly improve the mechanical properties of sample 9 over those of sample 6, even though sample 9 exhibited excellent weight retention at 315° C (600° F). These results indicate

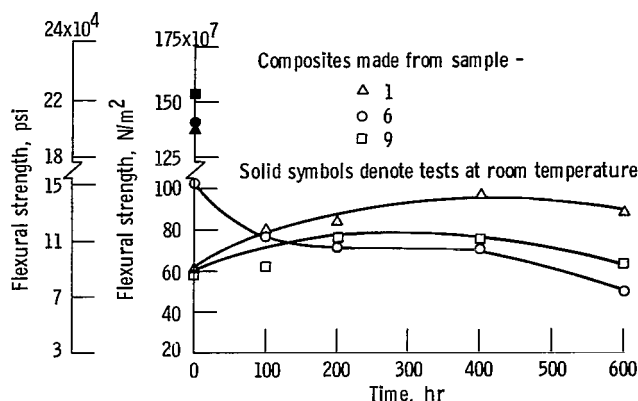


Figure 15. - Flexural strength of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

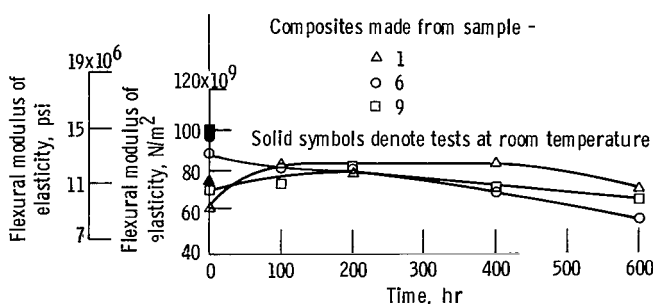


Figure 16. - Flexural modulus of elasticity of polyimide/HTS graphite fiber composites exposed and tested in air at 315° C (600° F). Specimens cut from 7.62 by 25.4 centimeter (3 by 10 in.) laminates.

that one cannot rely exclusively on isothermal weight loss data for prediction of composite performance.

CONCLUSIONS

Based on the results of the present investigation, the following conclusions can be drawn:

1. For addition-type polyimide/graphite fiber composites prepared by means of the monomeric reactants approach, the most suitable aromatic diamine was found to be 4,4'-methylenedianiline (MDA).
2. By replacing the dimethyl ester of pyromellitic acid (PMDE) with the dimethyl ester of 3,3',4,4'-benzophenonetetracarboxylic acid (BTDE), it was possible to fabricate addition-type polyimide/graphite fiber composites with better resin flow and improved

mechanical properties and thermooxidative stability in air at 315° C (600° F) after 600 hours than composites containing PMDE.

3. The monomer composition 5-norbornene-2,3-dicarboxylic acid monomethyl ester (NE) with MDA and BTDE at three different stoichiometric ratios yielded composites that possessed long-term mechanical properties and thermooxidative stability superior to those of P10P composites. The panels fabricated from the NE/MDA/BTDE formulation at a molar ratio of 2/3.09/2.09 (sample 1) gave the best overall mechanical properties of all panels studied. After a 600-hour exposure in air at 315° C (600° F), the interlaminar shear strength of sample 1 was 45 percent higher than that of P10P composites, and the flexural strength of sample 1 was 75 percent higher than that of P10P composites.

Lewis Research Center,
National Aeronautics and Space Administration,
Cleveland, Ohio, April 24, 1972,
134-03.

REFERENCES

1. Lubowitz, H. R.: Polyimide Polymers. U.S. Patent 3,528,950, Sept, 15, 1970.
2. Burns, E. A.; Lubowitz, H. R.; and Jones, J. F.: Investigation of Resin Systems for Improved Ablative Materials. Rep. TRW-05937-6019-RO-OO, TRW Systems Group (NASA CR-72460), Oct. 1, 1968.
3. Burns, E. A.; Jones, R. J.; Vaughan, R. W.; and Kendrick, W. P.: Thermally Stable Laminating Resins. Rep. TRW-11926-6013-RO-OO, TRW Systems Group (NASA CR-72633), Jan. 17, 1970.
4. Serafini, Tito T.; Delvigs, Peter; and Lightsey, George R.: Thermally Stable Polyimides from Solutions of Monomeric Reactants. NASA TN D-6611, 1972.
5. Walton, Henry M.: Potential Antimicrobial Agents. I. Alkyl 4-Oxo-2-Alkenoates. J Organic Chem., vol. 22, no. 3, Mar. 1957, pp. 308-312.
6. Bell, Vernon L.; and Jewell, Robert A.: Synthesis and Properties of Polyimidozopyrrolones. J. Polym. Sci., Part A-1, vol. 5, no. 12, 1967, pp. 3043-3060.
7. Hanson, M. P.; and Serafini, T. T.: Effects of Thermal and Environmental Exposure on the Mechanical Properties of Graphite/Polyimide Composites. NASA TN D-6604, 1971.



009 001 01 U 18 720818 500903DS
DEPT OF THE AIR FORCE
AF WEAPONS LAB (AFSC)
TECHNICAL LIBRARY/DCUL/
ATTN: E LCU BOWMAN, CHIEF
KIRTLAND AFB NM 87117

POSTMASTER: If Undeliverable (Section 158
Postal Manual) Do Not Return

"The aeronautical and space activities of the United States shall be conducted so as to contribute . . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

—NATIONAL AERONAUTICS AND SPACE ACT OF 1958

NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

TECHNICAL REPORTS: Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

TECHNICAL NOTES: Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

TECHNICAL MEMORANDUMS: Information receiving limited distribution because of preliminary data, security classification, or other reasons.

CONTRACTOR REPORTS: Scientific and technical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

TECHNICAL TRANSLATIONS: Information published in a foreign language considered to merit NASA distribution in English.

SPECIAL PUBLICATIONS: Information derived from or of value to NASA activities. Publications include conference proceedings, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

TECHNOLOGY UTILIZATION PUBLICATIONS: Information on technology used by NASA that may be of particular interest in commercial and other non-aerospace applications. Publications include Tech Briefs, Technology Utilization Reports and Technology Surveys.

Details on the availability of these publications may be obtained from:

SCIENTIFIC AND TECHNICAL INFORMATION OFFICE

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

Washington, D.C. 20546